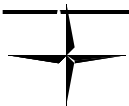

High Performance Opto-Mechanical and Structural Components in C/SiC-Technology (Short Carbon Fibre Reinforced Silicon Carbide)

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presented at

**NGST Technology Challenge Review 7/97
July 7-10, 1997
NASA Goddard Space Flight Center**



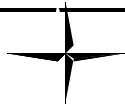
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Overview

- ☐ **C/SiC Material Technology**
- ☐ **Material Properties**
- ☐ **Applications**
- ☐ **Design Potential**
 - **Large Pieces**
 - **Coatings**
 - **A-thermal Structures**
 - **Mirror Design Concepts**
- ☐ **Conclusions**



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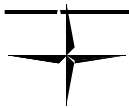


Conclusions

1. Material is Suitable for Ultra Lightweight Design of Opto-Mechanical Structures

- **Isotropic Characteristics**
- **CTE Compatibility to Coating Materials (SiC, Si, Glass)**
- **Easy Shaping of Complex Structures (Design Flexibility)**
- **No Shrinking**
- **Very Large Pieces by Joining**
- **No Open Porosity**
- **High Stiffness, High Strength**
- **Short Manufacturing Process**
- **Low Thermal Expansion at High Thermal Conductivity**

2. C/SiC is Suitable as Basic Material for Different Proposed NGST Mirror Designs



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Mirror Design Concepts

Passive Optics

**C/SiC Ultra Light Weight Substrate
with or without coating**

Active Optics

Monolithic Mirror

C/SiC Ultra Light Weight Substrate

Layer with Electrode Pattern

Layer from PZT/PMN Ceramics

Glass Layer (Polished like Monolithic Optics)

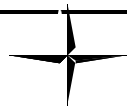
Stacked Facesheet Mirror

Support Structure Designed in C/SiC

Discrete PZT Stacks or Actuators

Glass Face

Both Mounted on Global Positioning Actuators on Deployable Structure



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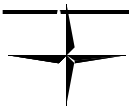
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Requirements

Advanced Composite Materials for Opto-Mechanical Applications:

- **Low Density**
- **Low Thermal Expansion**
- **High Stiffness**
- **High Strength**
- **Appropriate Thermal and Electrical Conductivity**
- **High Quality Optical Surfaces**
- **Flexible Design and Up-Scaling Capability**



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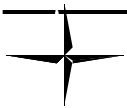
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Our Solution

C/SiC - Randomly Oriented Chopped Carbon Fibre Reinforced Silicon Carbide Structures with Polishible or Oxidation Protected Coatings Manufactured by:

- **Green Body Machining of C/C Raw Materials with Low Density and Randomly Orientated Chopped Carbon Fibre Reinforcement**
- **Infiltration Processing (without any Shrinkage) of C/C Structures with Pyrolytic Carbon and Liquid Silicon, Partly Reacted to SiC**
- **CVD-Deposition of Polishable or Oxidation Protective Coatings (e.g. Silicon Dioxid, Silicon Carbide, Glass) on the C/SiC Substrate Surface**
- **C/SiC Reveals a Variety of Optomechanical, Thermomechanical, Thermo-environmental and Configuration Aspects, which "Classical" Materials Like Metals, Glass or other Ceramics can not Provide Completely**
- **State of the Art Facilities Allow C/SiC Processing of Components with Diameters up to 3.0 Meter**



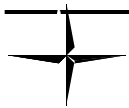
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C/SiC Material Characteristics/Properties (1)

- **Composition** SiC : Si : C 50-60 % : 20-30 % : 10-20 %
 (typical value)
- **Low Specific Density** (2,6 - 2,7 g/cm)
- **Tunable Stiffness** (240-260 GPa) and **Strength** (140-210 MPa)
- **Low Coefficient of Thermal Expansion [CTE]**, (20 °C-1000 °C
: 1,8- 4,1 x 10⁻⁶ K⁻¹)
- **High Thermal Conductivity and Diffusion** (~ 135 W/mK)
- **Low Electrical Resistance** (~ 200 x 10⁻⁶ Ohm x m)
- **Isotropic Material Properties and Characteristics**
- **Resistance to Corrosion/Oxidation and Abrasion**
- **High Hardness** (> 1500 N/mm)
- **High Temperature Resistance** (~ 2100 °C, Air), Optical Applications: ~ 1200 °C, Corrosive Applications: ~ 2100 °C



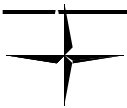
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C/SiC Material Characteristics/Properties (2)

- **High Thermal Shock Resistance (> 2100 K/sec)**
- **No Ageing and Creeping Effects (Stability of Shape)**
- **Non-magnetic**
- **Bio-compatible**
- **Damage Tolerance Behaviour**
- **Fast, Near Net Shape Infiltration Process Without Shrinkage**
- **Low-Cost Machining, no Special Tools Required**
- **Flexible Design and Large Structure Up-Scaling Capabilities**
- **High Quality Optical Surface Layers (e.g. SiC, Glass and Si, Roughness: < 5 Å)**
- **Optimized Application Design Capability Due to the Flexibility of Possible Variations of Material Process Parameters, etc.**



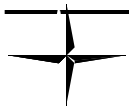
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Silicon Infiltration Processing

- Infiltration processing (without any shrinkage) of porous C/C-structures with molten silicon by capillary forces in high temperature vacuum process
- Densification and partial reaction of carbon matrix with silicon to silicon carbide (SiC)
- Processing temperature: 1500-2100 °C
- Processing pressure: < 100 mbar (vacuum)
- Reaction time: 15-60 minutes
- Short manufacturing process (24 hours from cold to cold)
- Near-net-shape infiltration process (minimal pre-grinding) without shrinkage and minimized stress introduction

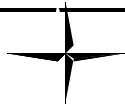
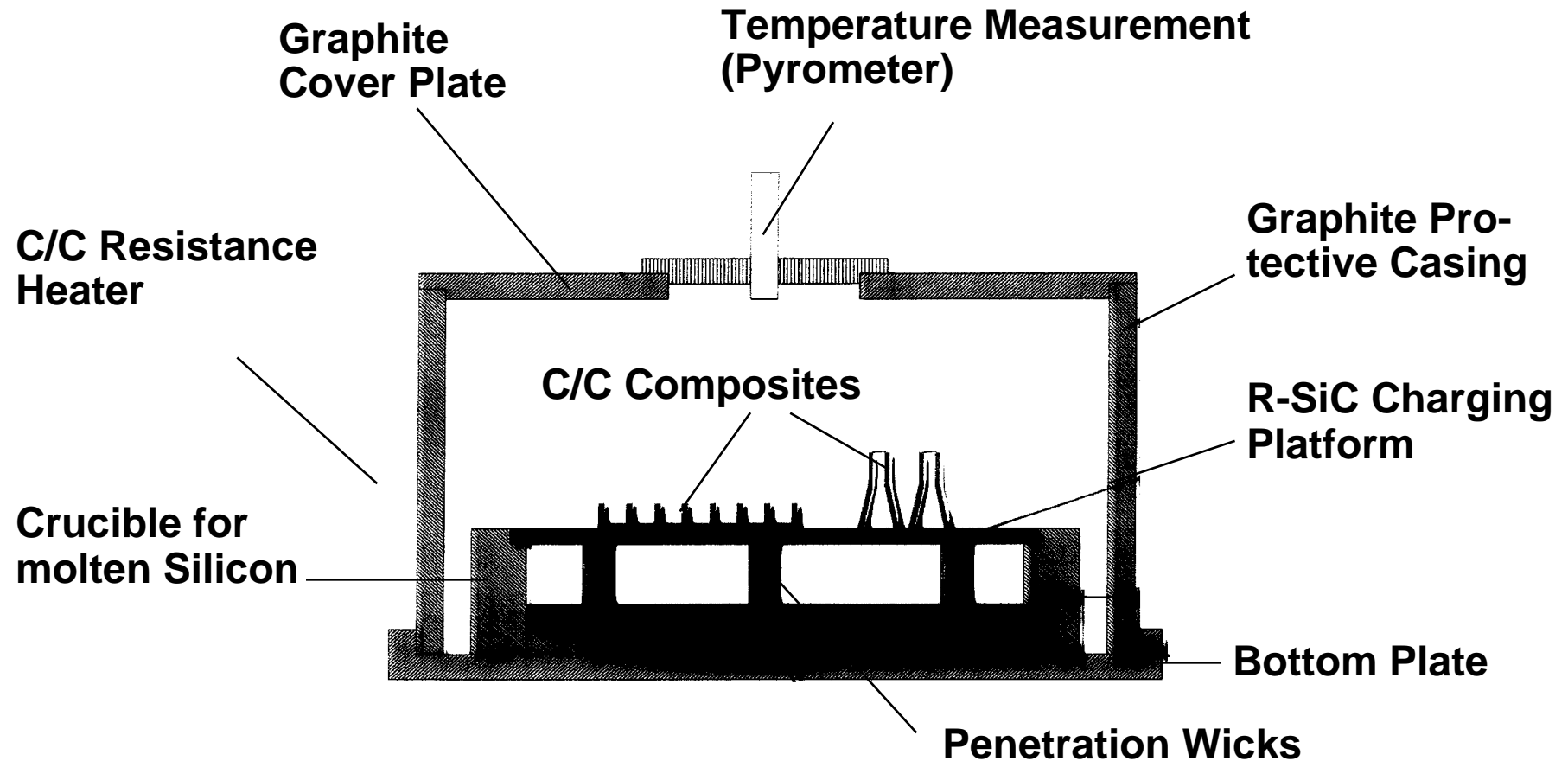


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Silicon Infiltration Process Facility

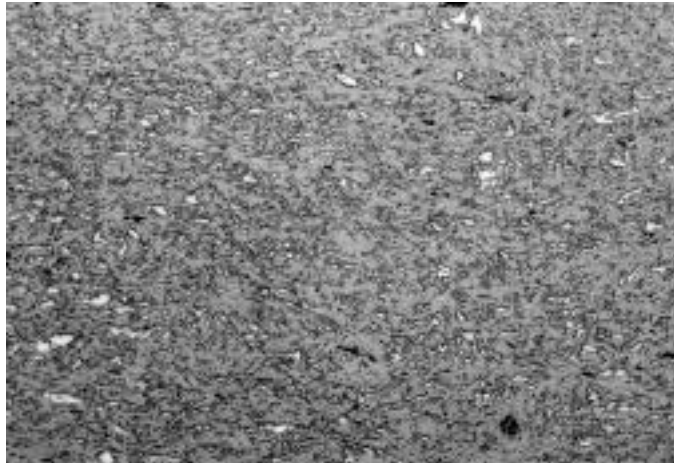


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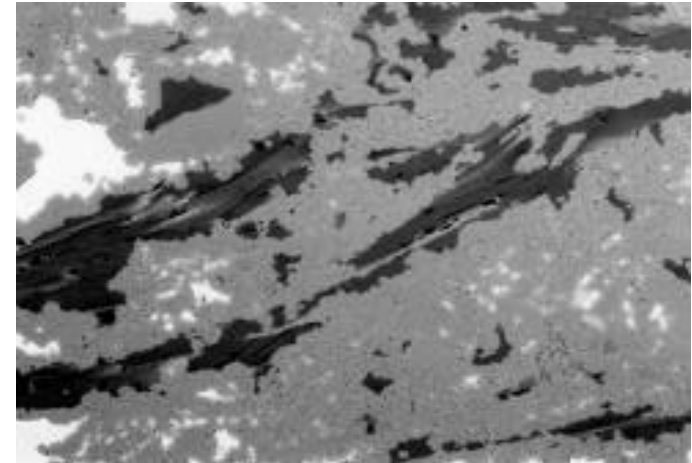
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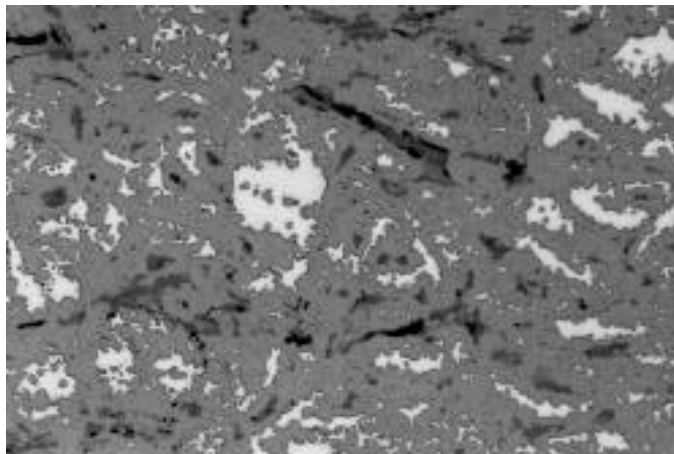
Material Structure after Infiltration



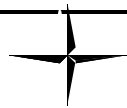
20 x



500 x



100 x



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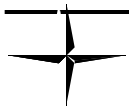
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Controlling & Tailoring of C/SiC-Material Properties

Optimized Application Design Capability due to Flexibility of possible Variations of Material Processes Parameters:

- **Carbon fibre/-matrix content of the C/C raw materials**
- **Density, pore volume and pore sizes of the C/C raw materials**
- **Carbon fibre type, -length, -pretreatment**
- **Carbon matrix forming process (reactivity of the carbon matrix)**
- **Carbon fibre protection systems (preventing chemical reactions, controlling fibre/matrix interface)**
- **Silicon infiltration process parameters (heat treatment, maximum temperatures, duration, pressure)**
- **Thermal treatments (graphitization), etc.**

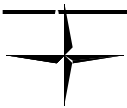
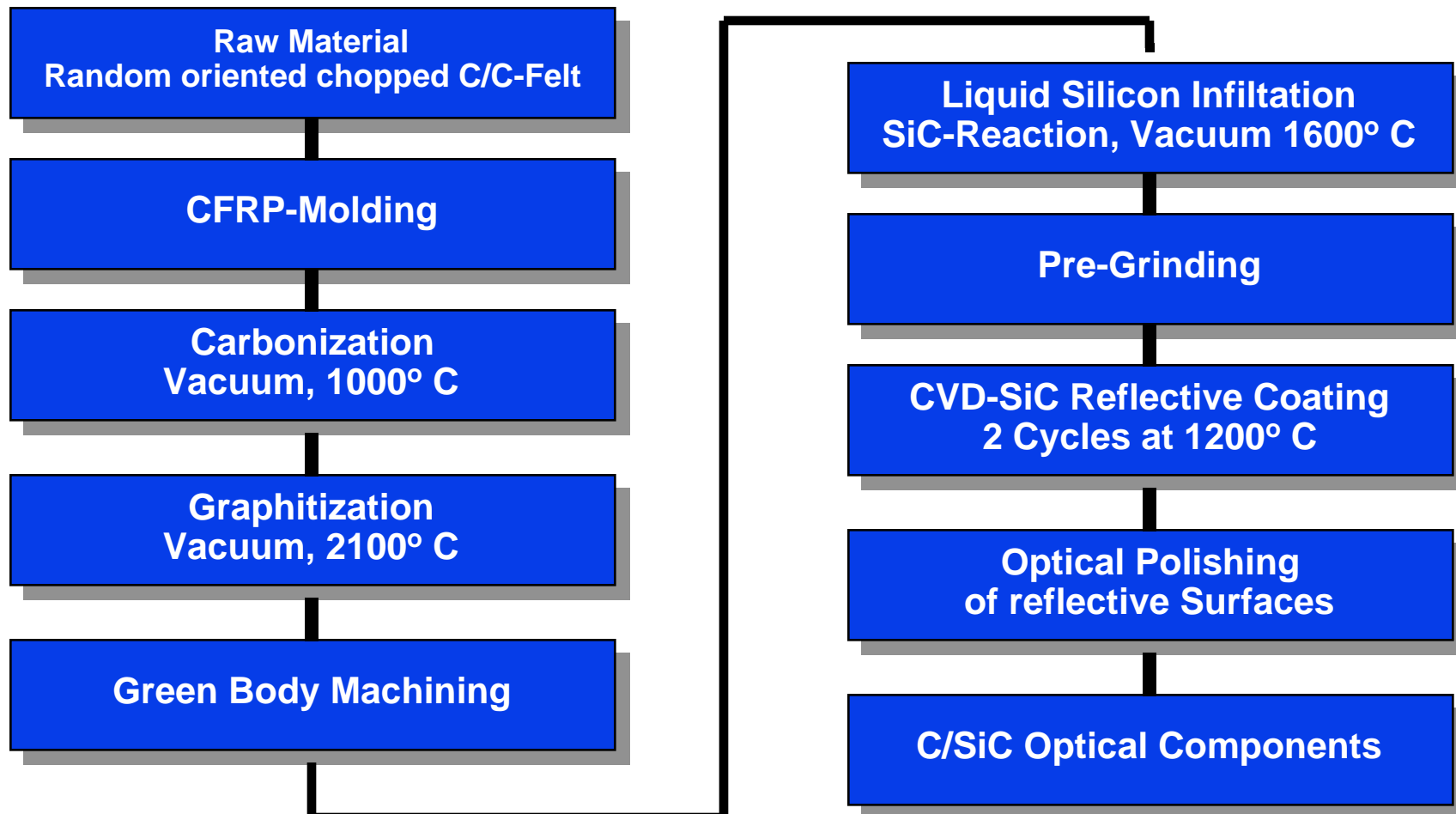


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Manufacturing Process of C/SiC-Mirror Structures

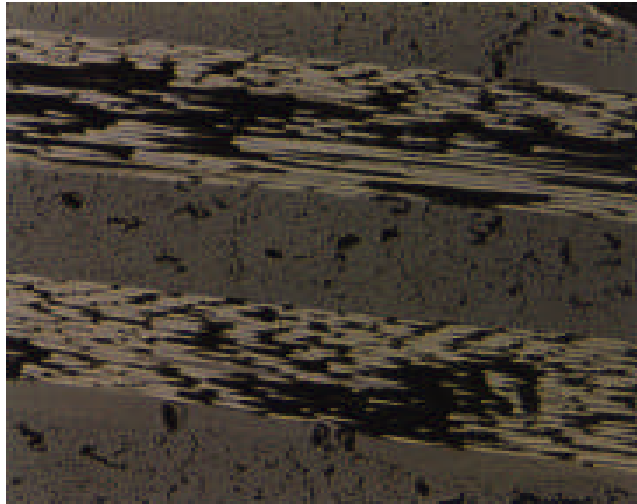


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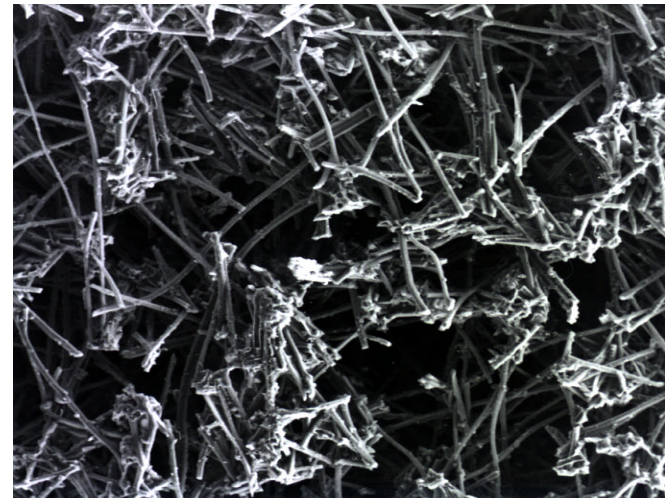
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CMC-Composites with Continuous (2-D) and Short Carbon Fibre Reinforcement

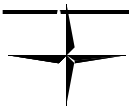


**Continuous (2-D)
Reinforcement**



**Short Carbon Fibre
Reinforcement**

- Carbon Fibres have an Anisotropical Behaviour Concerning Strength, Stiffness, CTE, Thermal Conductivity and Electrical Resistance
- Continuous Carbon Fibre Reinforcement leads to an anisotropical behaviour in the x-, y- and z-Direction of the CMC Composites
- Short Carbon Fibre Reinforcement (Randomly Oriented Leads to an Isotropical Behaviour in the x-, y- and z-Direction of CMC Composites)

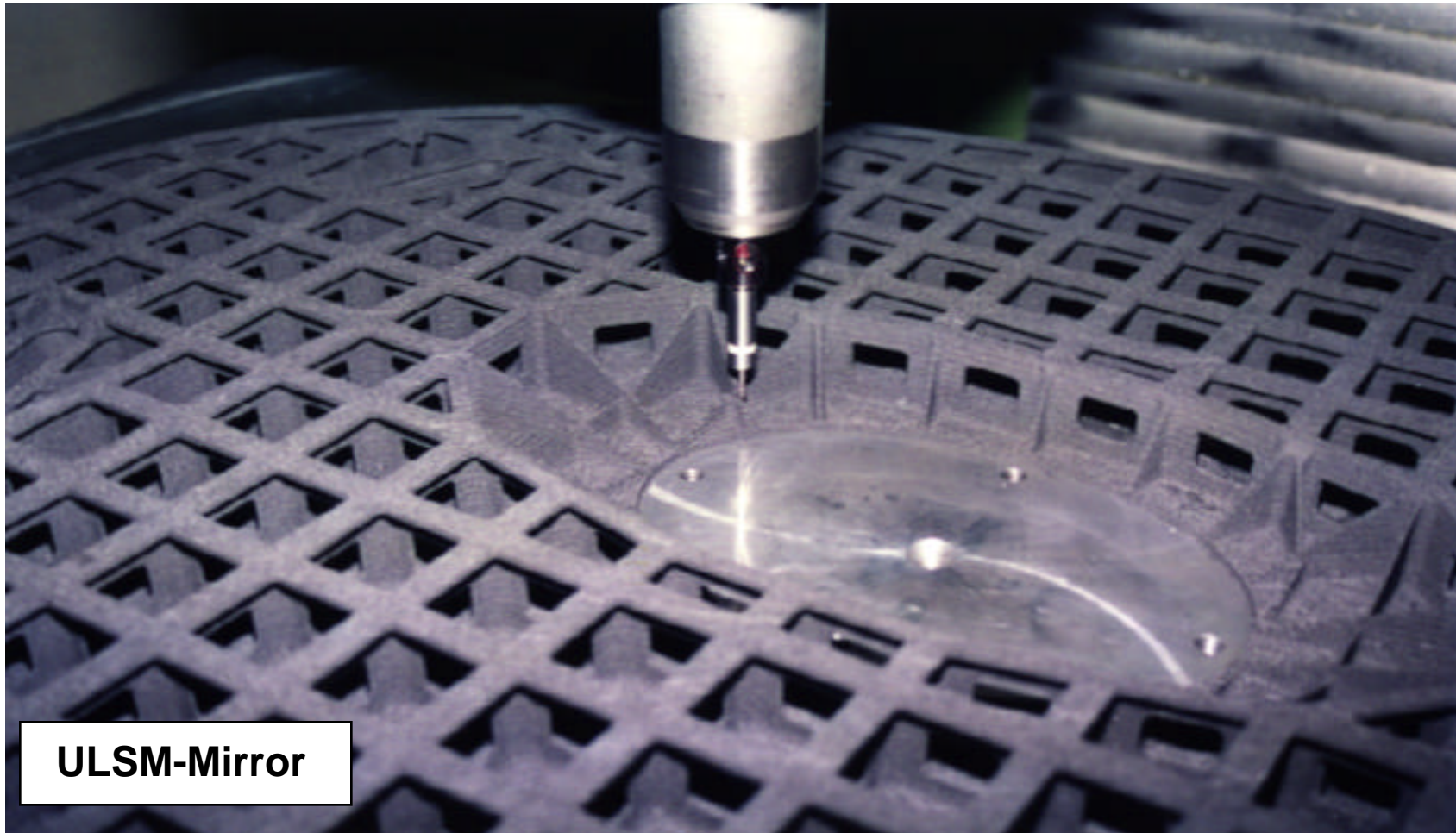


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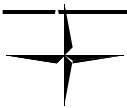
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Machining (Milling) of C/C-Felt (Green Body) with Conventional Processing Machines to Final Design



ULSM-Mirror



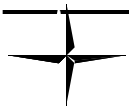
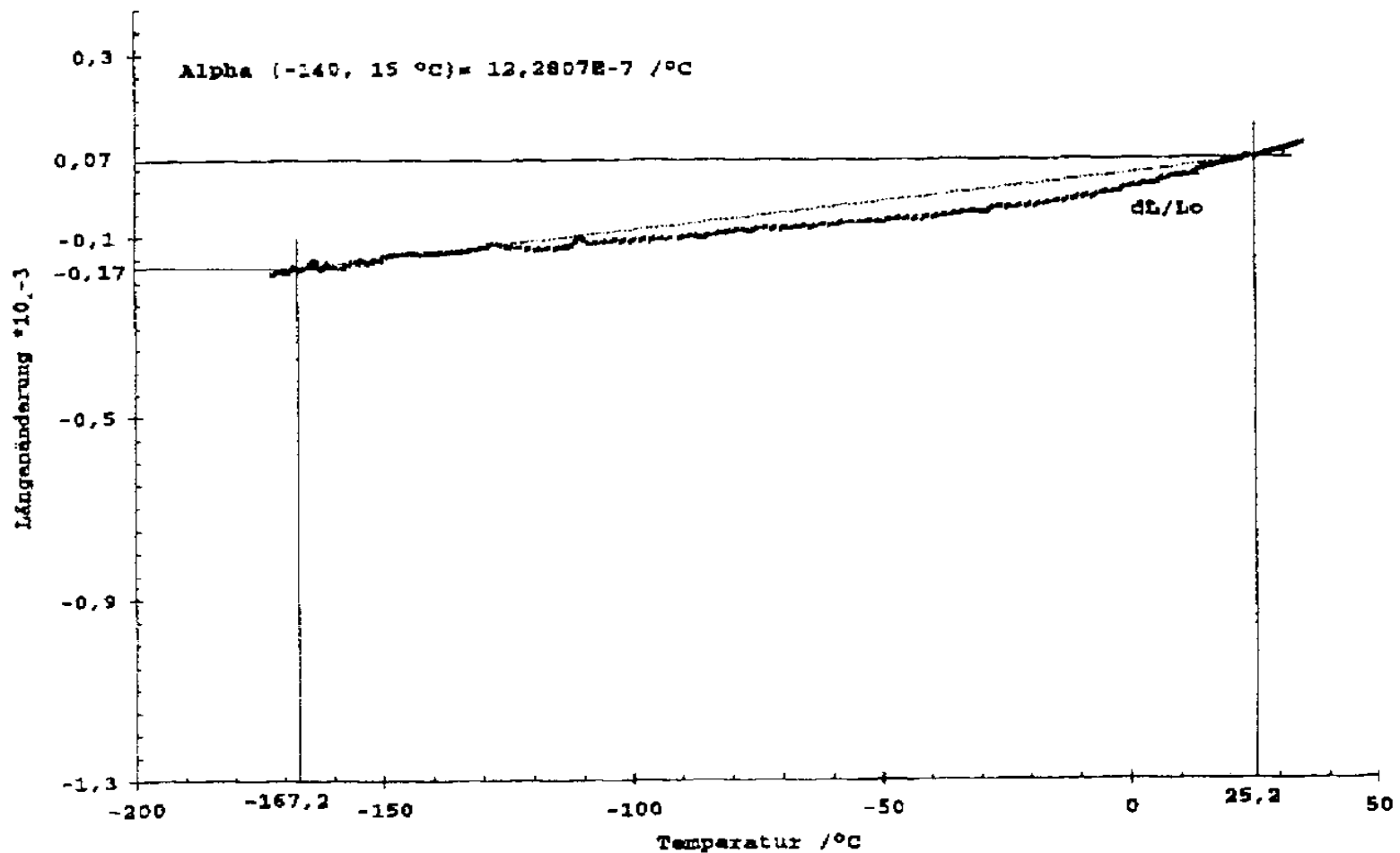
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CTE Behaviour of C/SiC

Range: - 167 °C (110 K) to + 25 °C (303 K)



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Tensile Tests With C/SiC at 20 °C, Stress Versus Strain

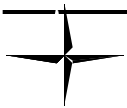
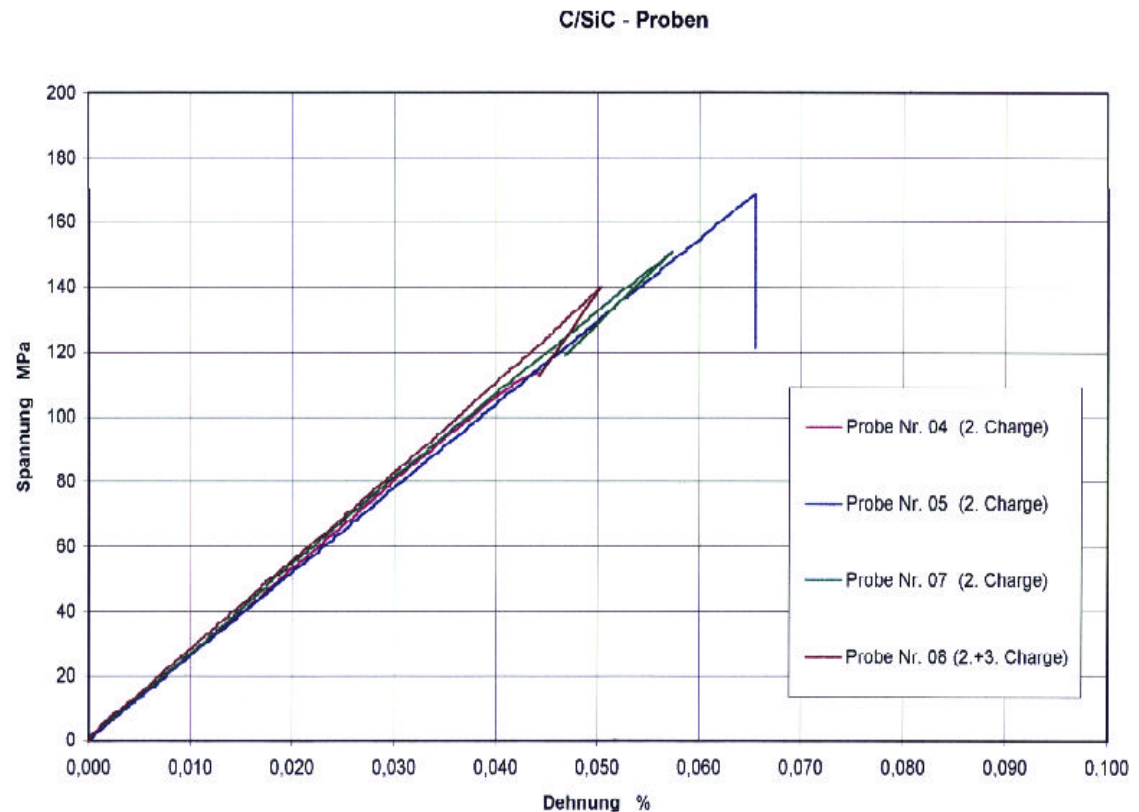
Density:
2.65 g/cm

Tensile Strength:
140 MPa

Tensile Modulus:
260 GPa

Bending Strength:
210 MPa

Bending Modulus:
235 GPa



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Tensile Tests With C/SiC at 20 °C, Stress Versus Strain

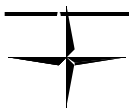
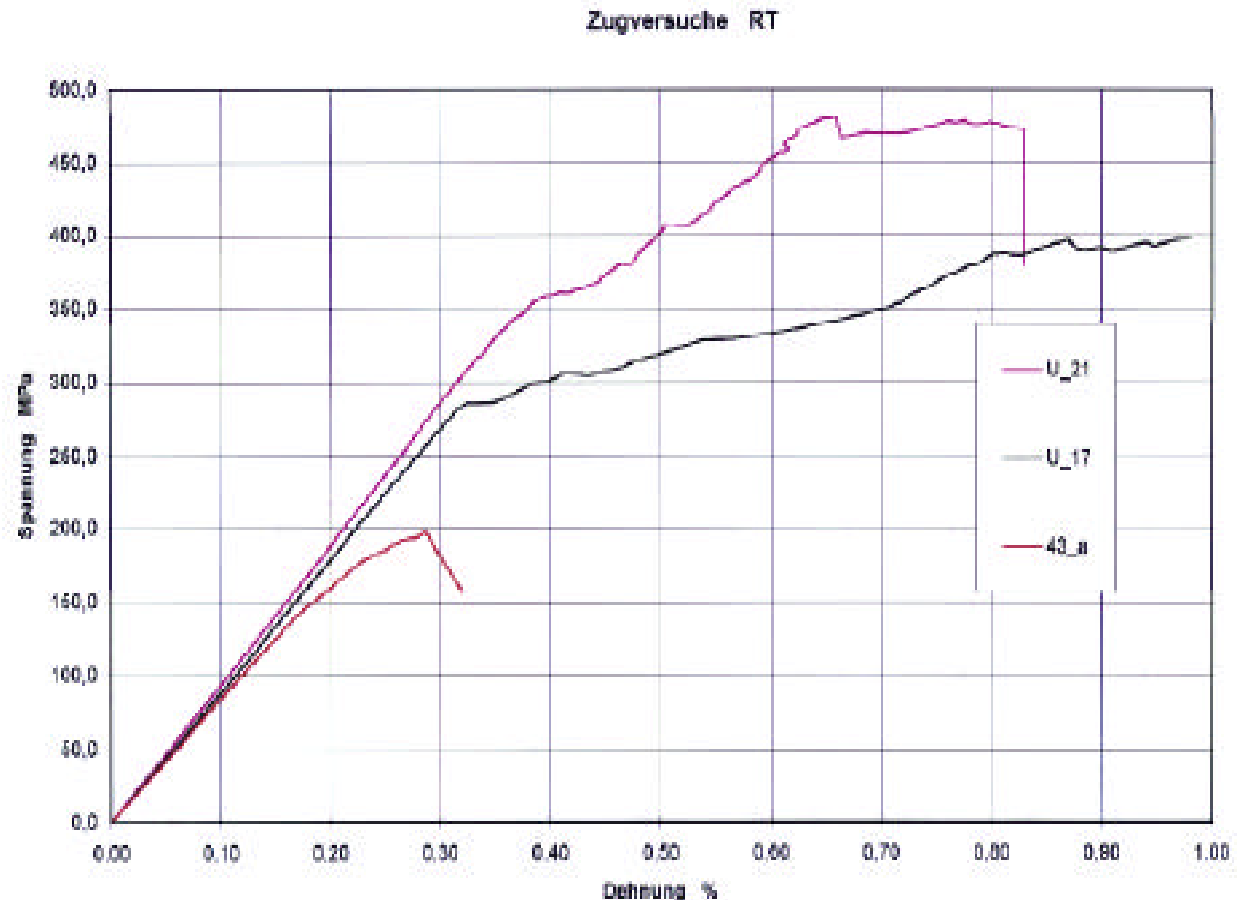
Density:
1,5 g/cm

Tensile Strength:
480 MPa

Tensile Modulus:
90 GPa

Bending Strength:
300 MPa

Bending Modulus:
85 GPa



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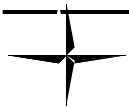
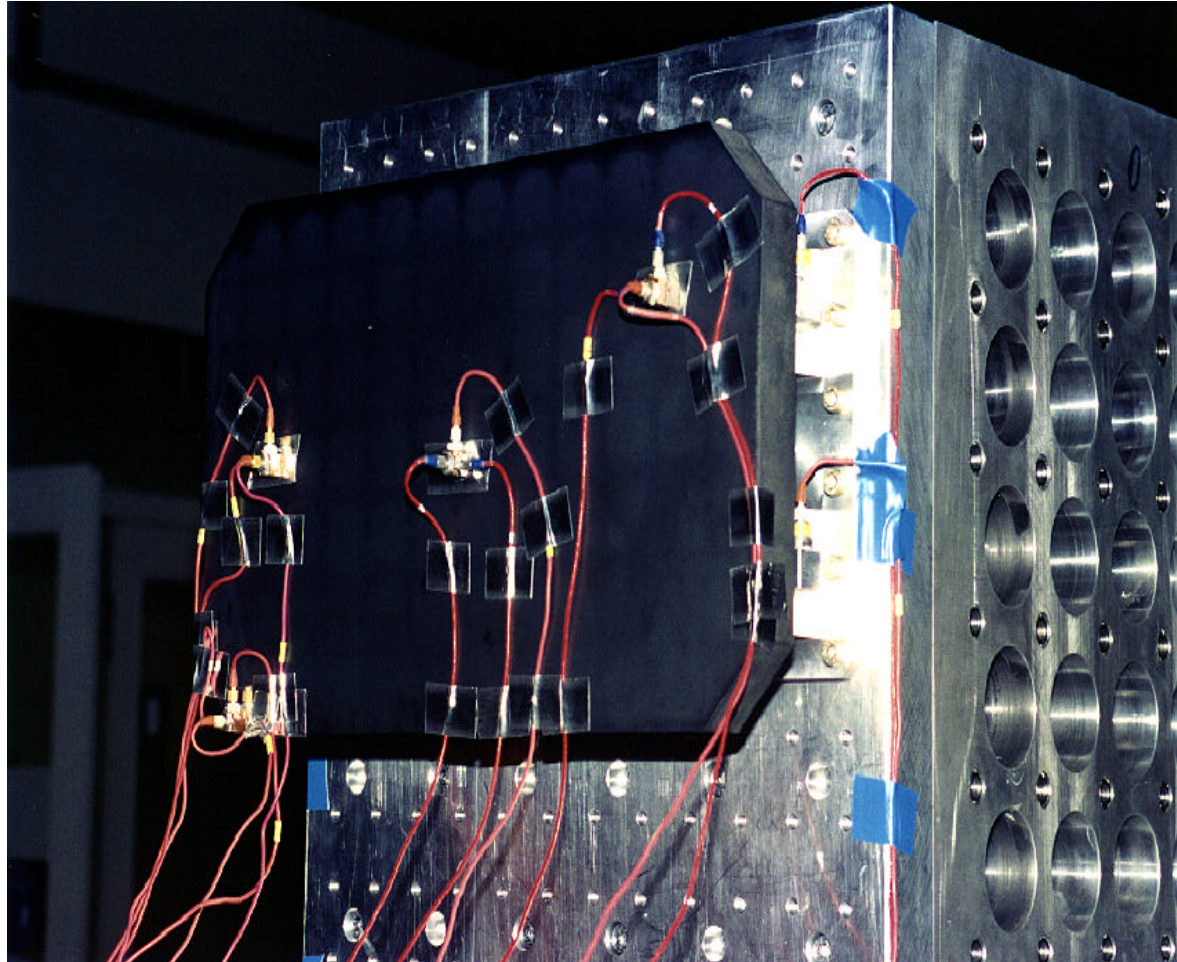


Shaker Vibration Test of Lightweight C/SiC Mirrors

45 g Loading in x-, y- and z-Direction

Mirror Dimension:
480 x 280 x 40 mm

Mirror Mass:
4 kg

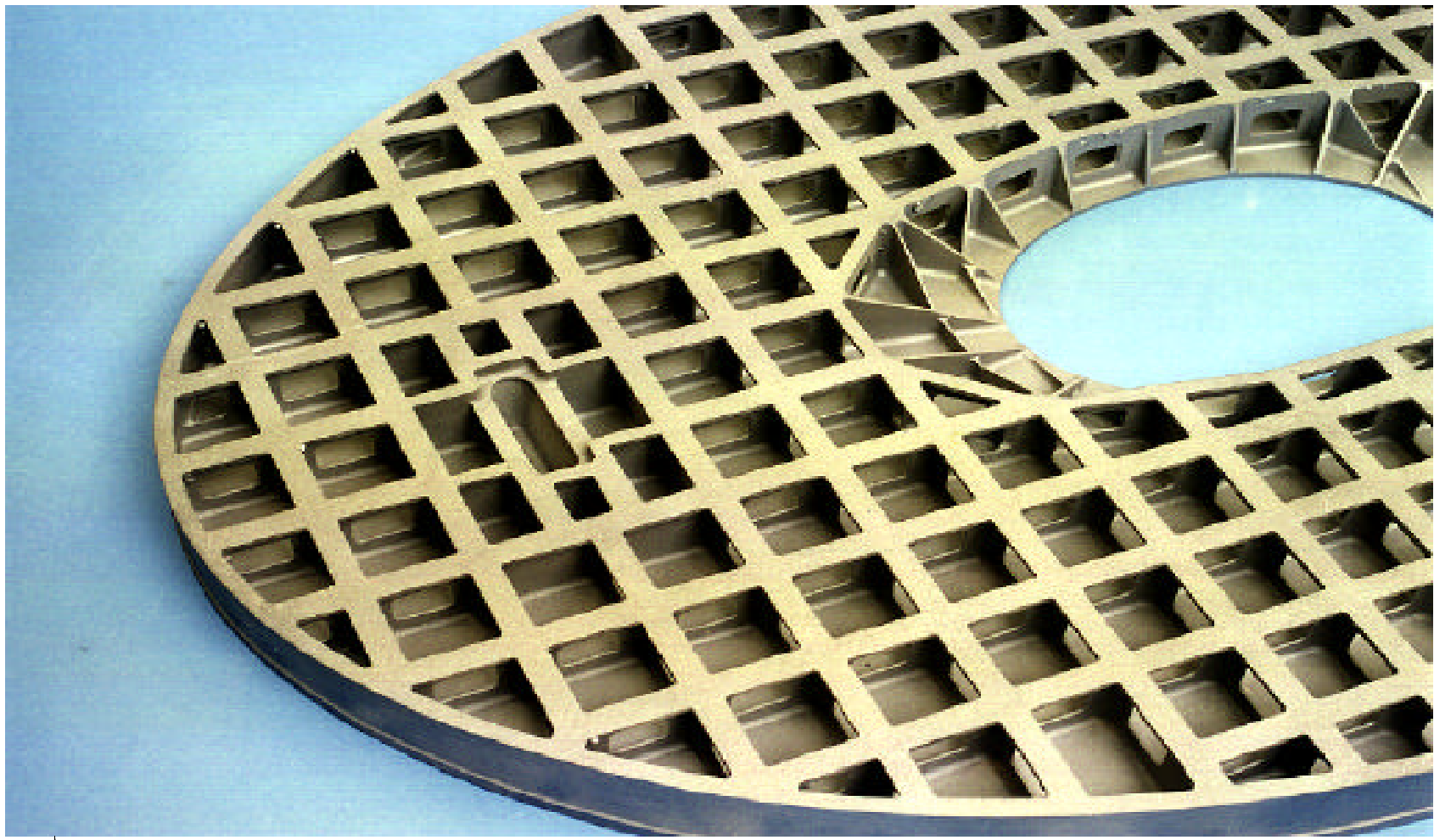


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Ultra-lightweight structure (rear side) of MSG C/SiC Scan Mirror



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C/SiC Scan Mirror Test Article

Dimensions:

480 x 280 x 40 mm

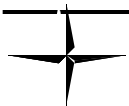
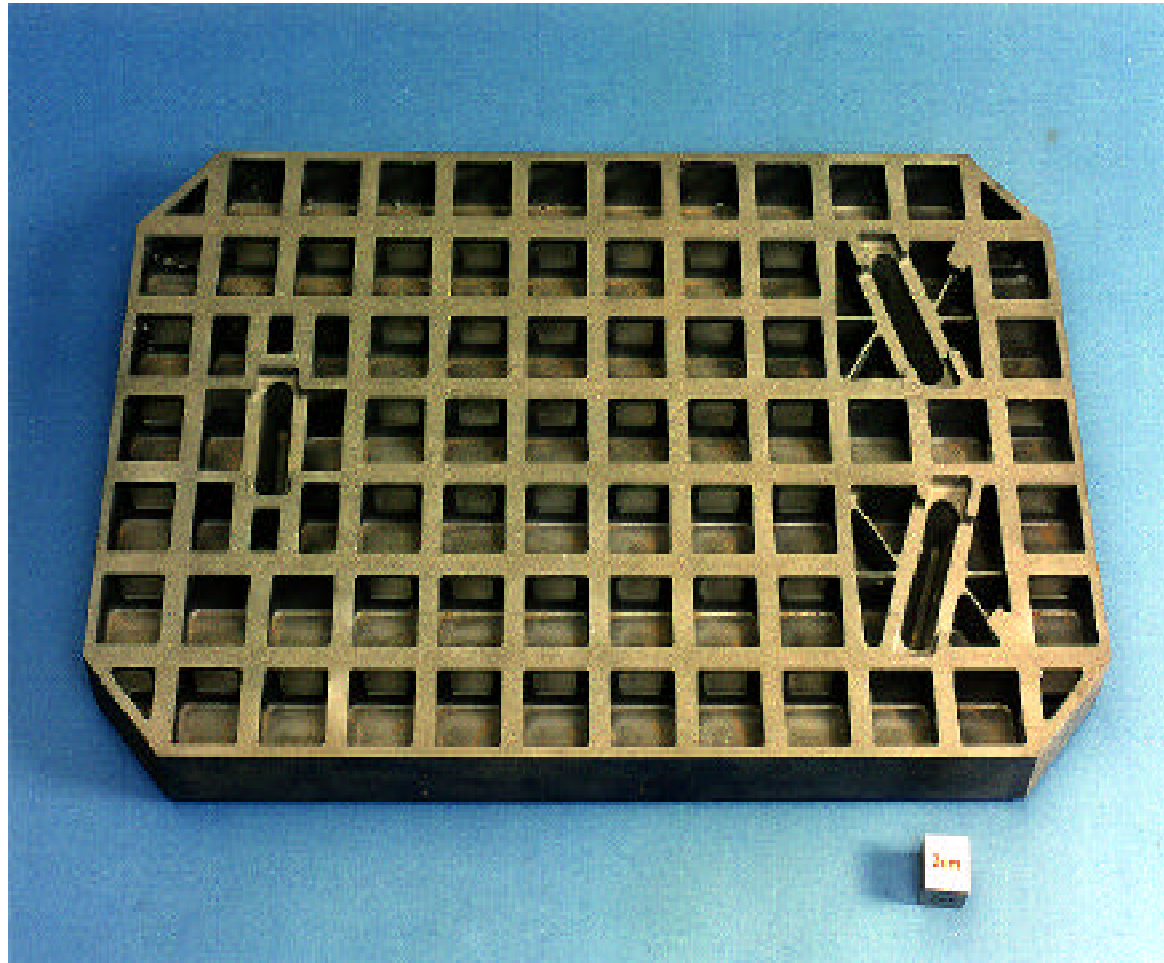
Mass: 3 kg

**First Eigenfrequency:
210 Hz**

Used for Testing:

- Thermal
- Mechanical
- Environmental

**With SiC-Coating on
Rear Side**

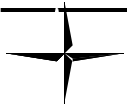
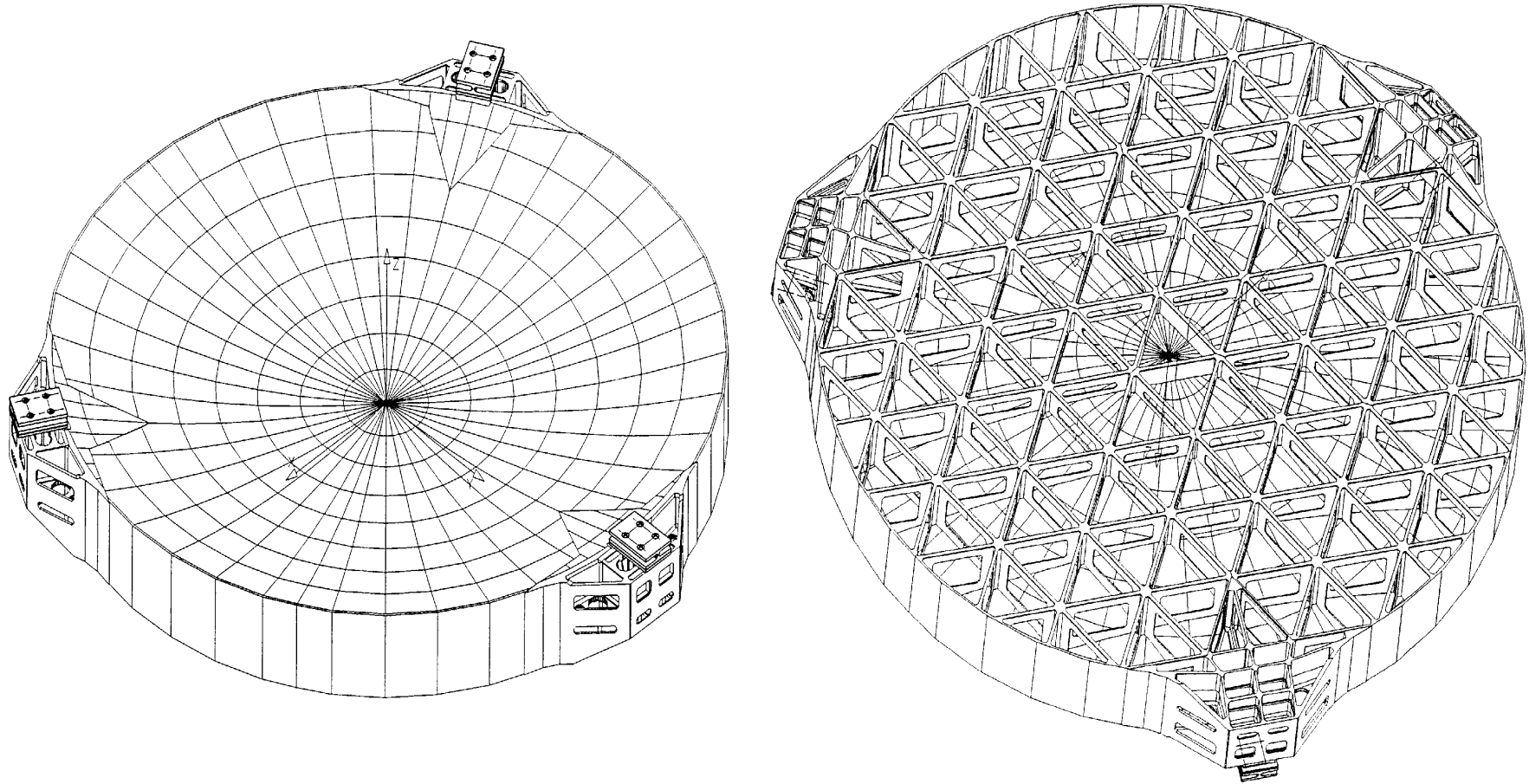


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ATLID Telescope Mirror Design Based on C/ SiC



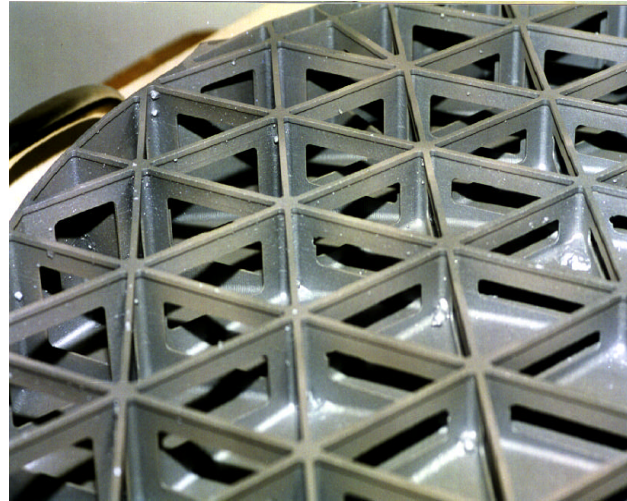
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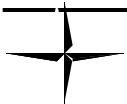
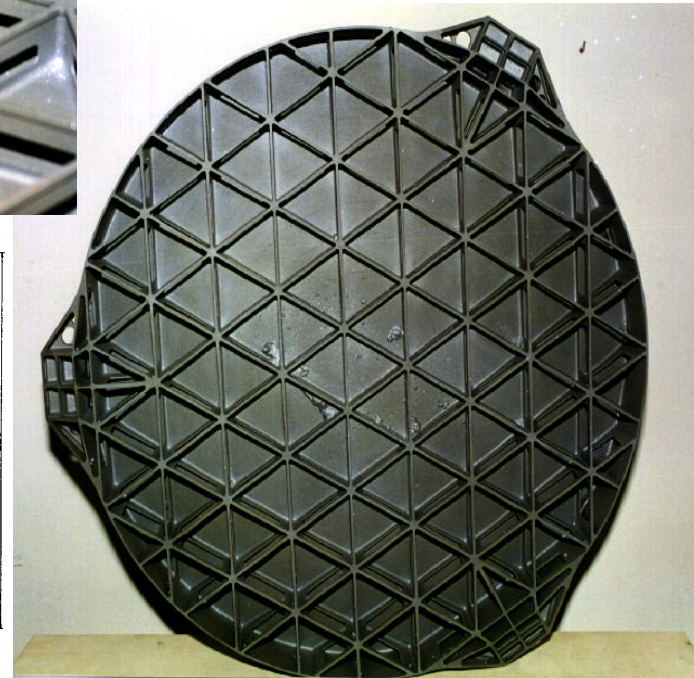
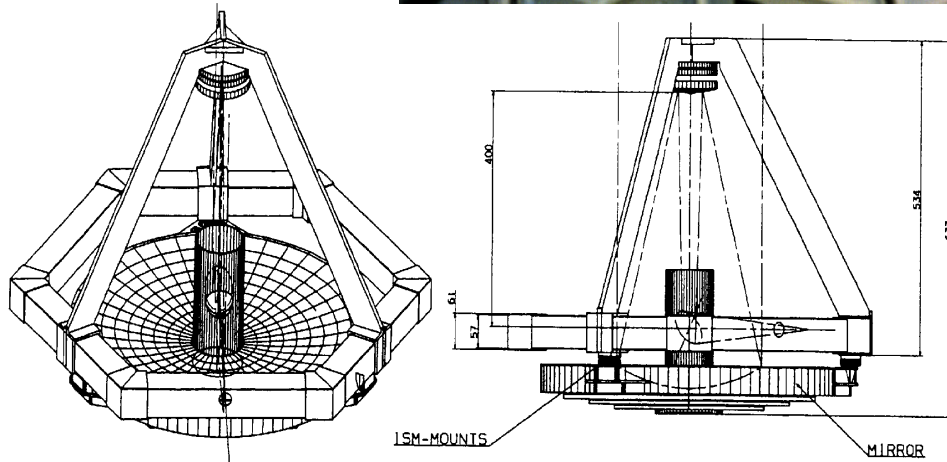


C/SiC ATLID Telescope Primary Mirror

Diameter: 630 mm
Mass: 6,0 kg (incl. coating and mounting prov.)
Rim Height: 70 mm
Rib Thickness: 1 mm
Parabolic $F=0,9$
Areal Weight =19,24 Kg/m^2



- Lightweighting 12 % (with resp. to solid dimensions)
- First Eigenfrequency 424 Hz
- Integrated Mounting Provision



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MSG C/SiC Scan Mirror

Dimensions:

805 x 520 x 40 mm

180 x 140 (center hole)

Mass: 8 kg

Areal Weight:

25,89 kg/m²

Micro-Roughness:

< 10 Angström

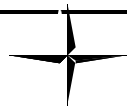
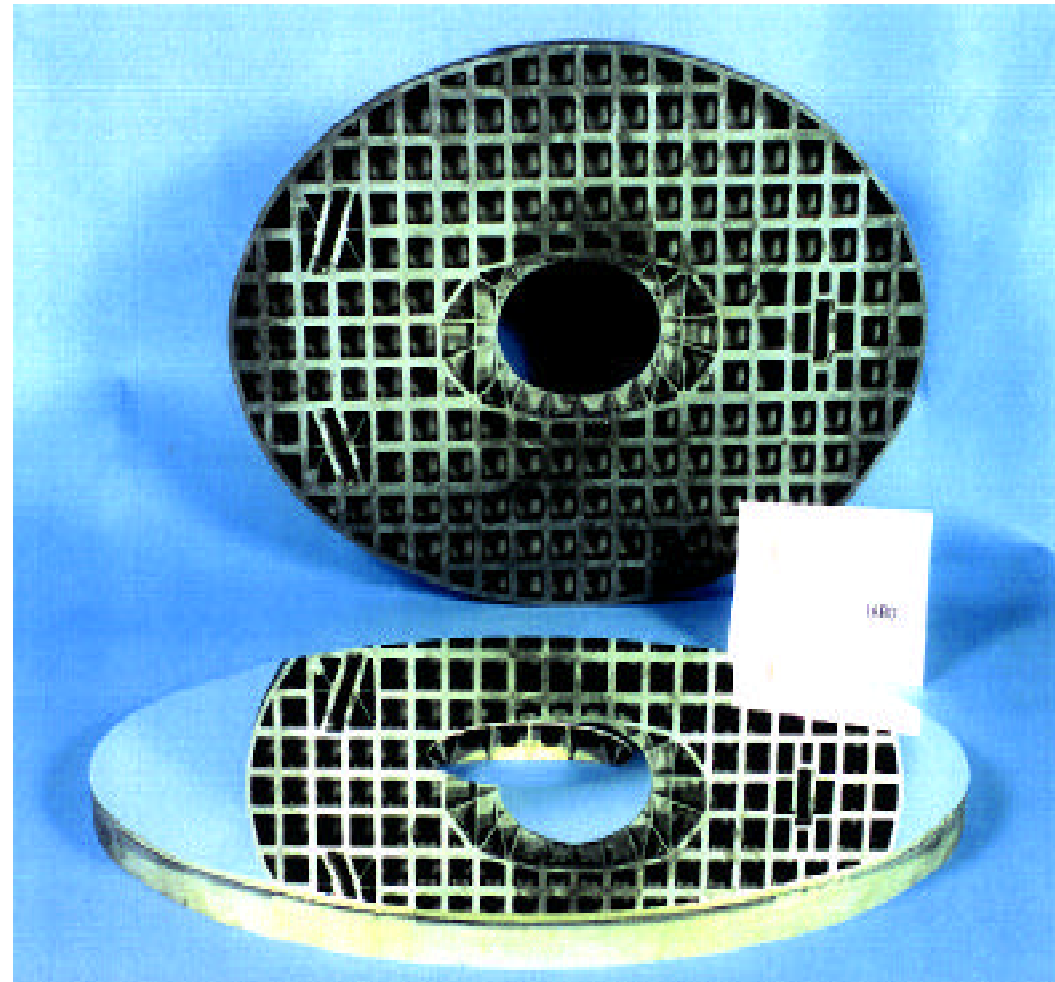
Optical Surface Quality:

$\lambda/6$

First Eigenfrequency:

400 Hz

Coated with CVD-SiC and Silver
Optically Polished
View from Front and
Rear Side



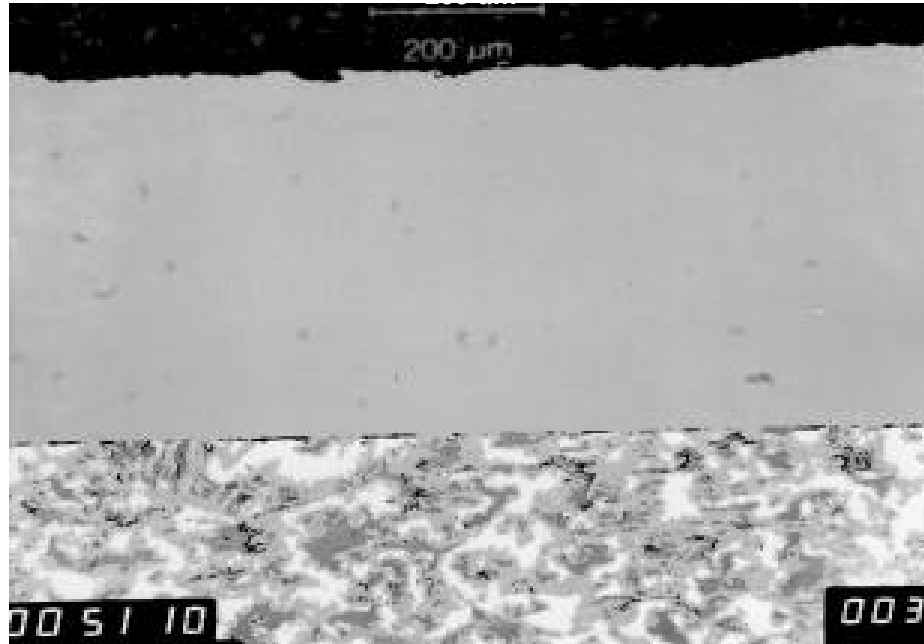
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Polishible Coating

- Si or SiC-Coating

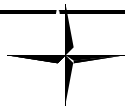


**CVD-SiC-Coating
of C/SiC**

**No Cracks due
to CTE-compa-
tibility**

- Glass Coating

**Hot Pressing of Glass Sheet on C/SiC Mirror
Substrate**

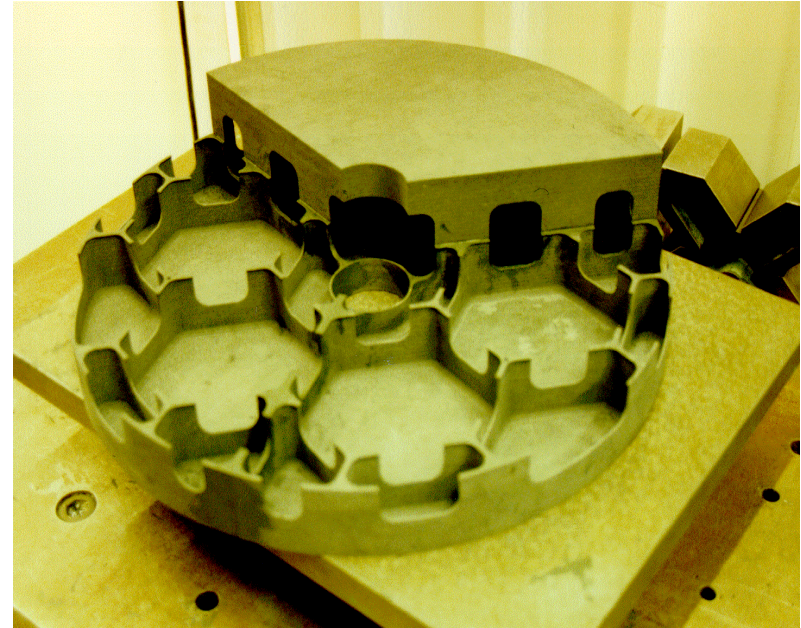
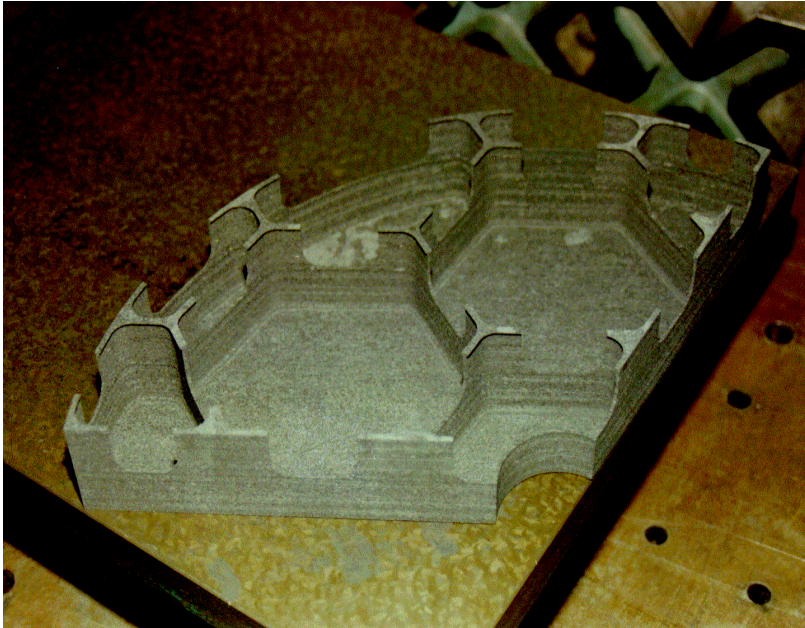


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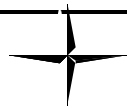
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Joining Techniques



- ☐ **Assembling of Parts (max. Size 200 x 200 x 20 cm) by Mechanical Fittings or Screws**
- ☐ **Fixation by Phenolic Fibre Resin Clou**
- ☐ **Converted to Carbon During Infiltration**
- ⇒ **Homogenous Material Texture at Joining Areas**

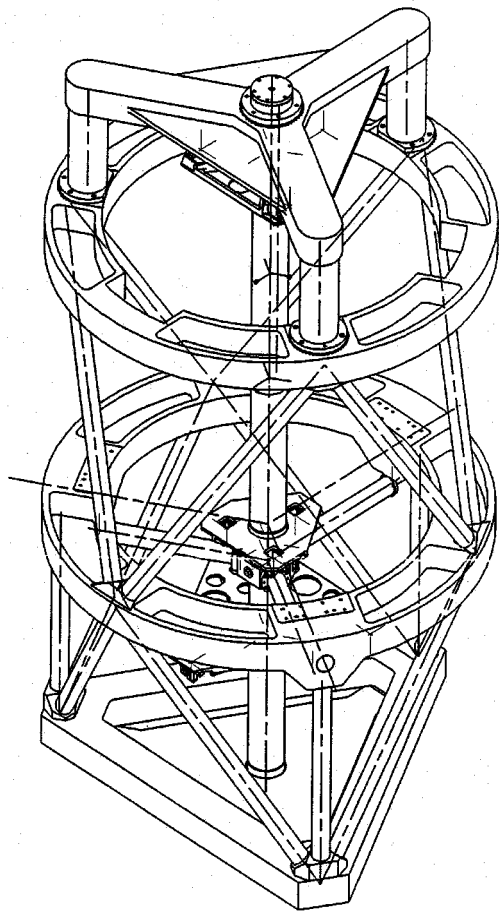


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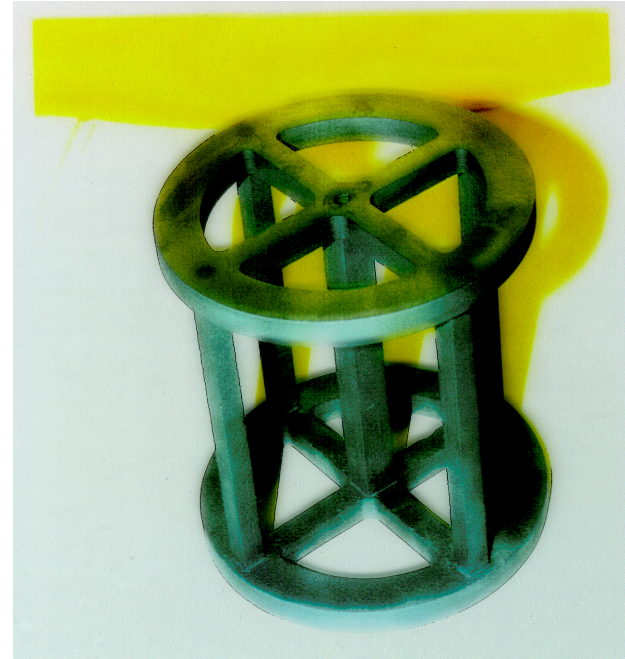
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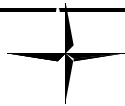
A-Thermal Design of Complex Telescope Structures



**Optical Bench
Actually in Manu-
facturing
Size:260X150 cm**



**Test Article for Lab Tests of
Stability During Thermal
Cycling**



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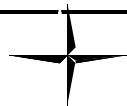
Thermal Performance

Thermal Figures of Merit:

			C/SiC	Zerodur	Be I-70A
CTE @ RT	α	10^{-6} K^{-1}	2,0	0,05	11
Thermal conductivity	k	W / m K	135	1,64	194
Specific heat	c	J / kg K	700	821	1820
Young's Modulus	E	GPa	260	90,6	289
Steady state thermal distortion	$E k / \alpha$		17.550	2.972	1.598
Dynamic thermal distortion	$E k / (\alpha c)$		25,07	3,62	2,80

Thermal Performance may be Influenced by Design and Manufacturing such as

- **Surface Treatment**
- **Lightweighting Ratio**
- **Seeing Factors**
- **Heat Conducting Cross Sections**



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